

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. REMOVED C-85-8 ENGINE AND INSTALLED CONTINENTAL O-200A48 SERIAL # 265429-R, USING 85 HP SHORT ENGINE MOUNT, ALSO HANLON WILSON EXHAUST SYSTEM (SAME AS CESSNA 150) USED STRUCTURAL MODIFICATIONS, INSTALLED USING PART NO.5 BA-809 L AND R FRONT SPAR STRUT FITTING B-809 L AND R REAR SPAR STRUT FITTING 10 5/8 FIBER BUSHINGS INSTALLED IN RIGHT AND LEFT FRONT SPARS AT BUTT FITTINGS
2. F-19 FUEL TANK (MAIN) WITH 1/4 OD RAM AIR VENT LINE INSTALLED, WITH AIR TIGHT GAS CAP ALL FUEL LINE CHANGED TO 3/8" AND LARGER GAS COILATOR ALSO INSTALLED
3. REMOVED SLING TYPE BAGGAGE AND INSTALLED 1/4" WOOD PANELING BAGGAGE COMPARTMENT SAME AS MODEL F-19, MAX BAGGAGE WT 30 LBS SEAPLANE, 50 LBS LANDPLANE
4. AIRCRAFT NEW WT AND BAL COMPUTATION
 EMPTY WT 975 LBS ON EDO 1320 FLOATS
 EMPTY WT C.G. 16.79 INCHES
 GROSS WT ON FLOATS 1351 LBS
 USEFUL LOAD 376 LBS
 GROSS WT SAME AS BC-65, ABOVE MODIFICATIONS MADE IN ACCORDANCE WITH TAYLORCRAFT INC. INSTRUCTIONS, SEE EXTRA SHEET

ADDITIONAL SHEETS ARE ATTACHED